



# UNIVERSITY OF RUHUNA

Faculty of Engineering

End-Semester 3 Examination in Engineering: July 2022

Module Number: ME3206

Module Name: Strength of Materials

[Three Hours]

[Answer all questions, each question carries 12 marks]

- Q1. a) Explain followings
- i Stress at a point
  - ii Normal Stress and Shear Stresses
  - iii Principal stress and principal planes
- [3.0 Marks]
- b) Consider a mechanical component under direct load  $F_1$  and  $F_2$  giving rise to stresses  $\sigma_y$  and  $\sigma_x$  vertically and horizontally as shown in Figure Q1.

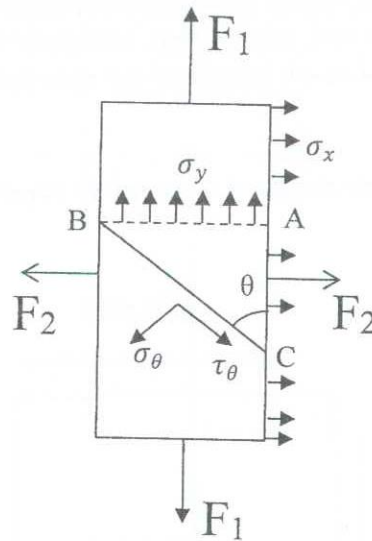


Figure Q1

Derive equations to find  $\tau_\theta$  (shear stress along BC plane) and  $\sigma_\theta$  (normal stress to BC plane).

- c) By examining the equations, conclude the followings;
- i Maximum normal stress
  - ii Maximum shear stress

[2.0 Marks]

- d) A cast iron block of  $10 \text{ cm}^2$  cross-section carries an axial tensile load of  $50 \text{ kN}$ . Calculate;
- Magnitude of the normal stress and shear stress on a plane inclined at  $30^\circ$  to the axis of the block
  - Maximum shear stress in the block

[4.0 Marks]

- Q2. a) Derive an expression for the theoretical fracture strength of a material. [4.0 Marks]
- b) Modulus of elasticity of steel is  $25 \times 10^{10} \text{ Pa}$ . Calculate the theoretical fracture strength of steel. [2.0 Marks]
- c) In actual fracture strength of steel is  $22.7 \times 10^9 \text{ Pa}$ . Is there any difference between actual and theoretical fracture strength values of steel?. If it is so, what are the reasons for that? [2.0 Marks]
- d) What do you understand by Stress Intensity Factor and Critical Stress Intensity Factor (Fracture toughness)? (Clue-  $K_I = \sigma \sqrt{\pi a} f(\alpha, a)$ ) [2.0 Marks]
- e) Suppose that a wind component on an aircraft is fabricated from an aluminum alloy. During an experiment, it has been determined that a fracture occurs at a stress of  $450 \text{ MPa}$  when the maximum internal crack length is  $2.6 \text{ mm}$ . For this same component and same alloy, compute the stress level at which fracture will occur for a critical crack length of  $4.8 \text{ mm}$  (internal). [2.0 Marks]

- Q3. a) The state of plane stress at a point is represented by the stress element below (Figure Q3).

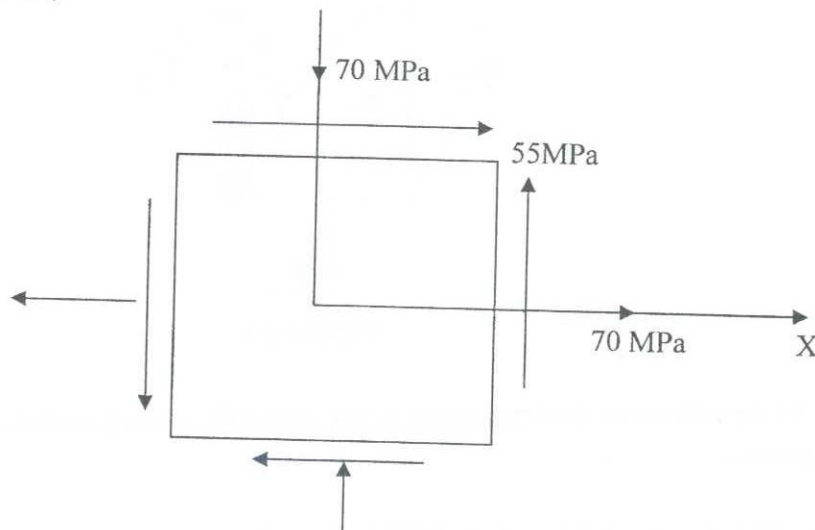


Figure Q3

Draw the corresponding Mohr's stress circle for the above stress system.

[3.0 Marks]

- b) Find the stresses on an element inclined at  $230^\circ$  clockwise and draw the corresponding stress elements.

[3.0 Marks]

c) Calculate (or find using corresponding Mohr's stress circle) the principal stresses and the planes on which they act. [3.0 Marks]

d) What would be the effect on above results if owing to a change of tensile loading to compressive while shear stresses remain unchanged? [3.0 Marks]

Q4. a) What do you understand by 'fatigue' failure of materials? [2.0 Marks]

b) What are the stages of fatigue failure and explain them briefly with suitable sketches. [3.0 Marks]

c) What are the roles of the following kinds of stress in fatigue failure?  
 i Compressive stress  
 ii Tensile stress [2.0 Marks]

d) State the Miner's law of cumulative fatigue. [2.0 Marks]

e) Figure Q4 shows a part of S-N curve of a metallic material. Rotating shaft of a machine was fabricated by above mentioned metal. When it was used following rotations had been done.

Stress (MPa)	Number of cycles operated (Under particular stress)
400	$4 \times 10^3$
160	$1 \times 10^2$

Due to a new requirement, operating stress is changed to 240 MPa. Then how many further rotations can it be operated before failure? [3.0 Marks]

Q5. a) What is the Griffith's criterion for fracture of brittle materials? [2.0 Marks]

b) Using the above mentioned (Q5. a)) criterion show the fracture strength of a material,

$$\sigma_f = \sqrt{\frac{8E\gamma_s}{\pi a(1+\nu)(1+k)}}$$

Where,

- $\sigma_f$  - Fracture strength of a material
- $E$  - Modulus of elasticity and
- $\gamma_s$  - Specific surface energy of a fractured surface.
- $\nu$  - Poisson's ratio
- $a$  - Half-length of a crack

If

$$k = \frac{3-\nu}{1+\nu}$$

derive an expression for fracture strength for plane stress conditions.

- [2.0 Marks]
- c) Explain briefly three (03) reasons for occurrence of fracture in materials. [3.0 Marks]
- d) Explain how the following factors cause a material lead towards one type of fracture (Ductile or Brittle fracture). [3.0 Marks]
- Temperature
  - Dislocation density
  - Grain size
- e) Describe all the steps involve with ductile fracture initiation and propagation. [2.0 Marks]

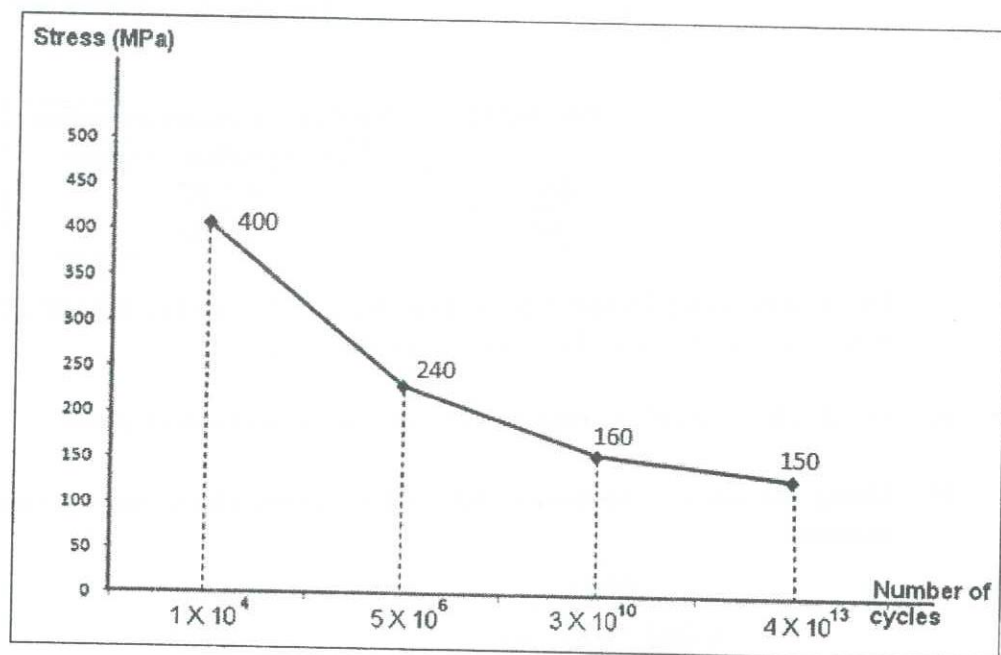


Figure Q4